

ORIGINAL

Code 015
20 SEP 1967

SPECIAL HANDLING REQUIRED IN ACCORDANCE WITH OPNAVINST 3750.6 SERIES

THIRD ENDORSEMENT on TRARON TWO, accident, ser 1-68A, concerning T-28C,
BuNo 140655, of 8 Aug 1967, pilot KAUFMAN

From: Chief of Naval Air Training
To: Commander, Naval Aviation Safety Center

Subj: Aircraft accident report; forwarding of

1. Forwarded, concurring in the conclusions and recommendations of the
Aircraft Accident Board and the action taken in the second endorsement.

F. J. Moore, Jr.
F. T. MOORE, Jr.
Chief of Staff

Copy to:
COMNAVAIRSYSCOM
DIRAFIP
CNABATRA
COMNAVPLANTREPO COLUMBUS
CO TRARON TWO

ORIGINAL

Code 015
11 SEP 1967
OPNAVINST 3750.6

**SPECIAL HANDLING REQUIRED IN ACCORDANCE WITH OPNAVINST 3750.6
SERIES**

**SECOND ENDORSEMENT on TRARON TWO AAR Ser 1-68A, concerning
T-28C, BuNo 140655, occurring 8 August 1967, Pilot: KAUFMAN**

**From: Chief of Naval Air Basic Training
To: Commander, U. S. Naval Aviation Safety Center
Via: Chief of Naval Air Training**

Subj: Aircraft Accident Report; forwarding of

**1. Forwarded concurring with the conclusions and recommenda-
tions of the Aircraft Accident Board.**

(b) (5)



**Copy to:
NAVAVNSAFCE (2 direct)
NAVAIRSYSCOMHQ
NAVPLANTREPO Columbus
TRARON TWO
Director AFIP**


D. H. GUINN

1 September 1967

SPECIAL HANDLING REQUIRED IN ACCORDANCE WITH OPNAVINST 3750.6 SERIES

FIRST ENDORSEMENT on TRARON TWO AAR 1-68A, concerning T-28C, 140655, occurring 8 August 1967, Pilot Kaufman.

From: Commanding Officer, Training Squadron TWO
To: Commander, U. S. Naval Aviation Safety Center

Via: (1) CNABATRA
(2) CNATRA

Subj: Training Squadron TWO AAR 1-68A, forwarding of

1. Forwarded concurring with the findings, conclusions and recommendations of the Aircraft Accident Board.



A. B. DAVIS

PART 1 GENERAL

1. AIRCRAFT ACCIDENT BOARD APPOINTED BY C.O. TRARON TWO	2. SERIAL NO. 1-68A	3. DTG (LOCAL) OF MISHAP 082240 R	4. MODEL AIRCRAFT T-28C	5. BUREAU NUMBER 140655
6. TO: Commander, Naval Aviation Safety Center		9. LOCATION OF MISHAP 1 MILE NORTH NAAS WHITING FLD ALFA		10. DAMAGE 3D1
7. VIA: CNABATRA	8. R/R N/A	11. TIME OF DAY NIGHT	12. TIME IN FLIGHT 0+02	13. FLIGHT CODE 3D1
14. CLEARED		15. TYPE CLEARANCE VFR LOCAL		
16. FROM: NAAS WHITING FLD		17. TO: NAAS WHITING FLD		
18. AMS/FEET 180 (EST)		19. A/C WEIGHT 7862		
12. BRIEF DESCRIPTION OF MISHAP UNCONTROLLED COLLISION WITH GROUND		15. ELEVATION AT TIME OF MISHAP S.L. 200'		
20. LIST MODEL, BUND, REPORTING CUSTODIAN AND DAMAGE CLASSIFICATION OF ANY OTHER A/C INVOLVED (Complete OPNAV Form 3750-1 for each A/C)		TERRAIN 0		
NONE				



1. NAME (Last, first, & middle initial) KAUFMAN, MARK R.	2. GRADE ENS	3. SERVICE NO. (b) (6)	4. TYPE OF SERVICE USNR	5. AGE 21	6. TOTAL A/C HRS 0	7. GRADE STUDENT	8. POSITION F/C	9. SIGNATURE ALFA
10. CO-PILOT (Identify & submit separate page 2)								
11. ALL MODELS		135		17. CV LANDINGS DAY/NIGHT		ALL 0 / 0		
12. ALL MODELS IN LAST 12 MONTHS		135		18. FCLP LANDINGS LAST 6 MONTHS DAY/NIGHT		ALL 0 / 0		
13. ALL MODELS IN LAST 3 MONTHS		59		19. INSTRUMENT HOURS LAST 3 MONTHS ACTUAL/SIMULATED		ALL 17 / 17		
14. ALL SERIES THIS MODEL		107		20. NIGHT HOURS LAST 3 MONTHS		ALL 9 / 9		
15. ALL SERIES THIS MODEL LAST 12 MONTHS		107		21. TOTAL HOURS IN JETS (if jet mishap) HELOS (if helo mishap)		0		
16. ALL SERIES THIS MODEL LAST 3 MONTHS		59		22. LAST PRIOR FLIGHT ALL SERIES THIS MODEL		DATE 3 AUG 1967		
23. DATE/GRADE LAST NITOPS START/PROXIMATION CHECK		N/A		24. TYPE INSTRUMENT CARD		NONE		
25. NAME (Last, first, & middle initial)								
NONE								

PART II MAINTENANCE MATERIAL AND FACILITIES DATA										
A. A/C HISTORY	1. DATE OF MANUFACTURE	2. FLIGHT HRS. SINCE ACCEPTANCE	3. NO. OF PAR/OVERHAUL	4. MONTHS SINCE LAST PAR/OVERHAUL	5. FLT. HRS SINCE LAST PAR/OVERHAUL	6. LAST PAR/OVERHAUL ACTIVITY	7. TYPE OF LAST CHECK PERFORMED	8. FLIGHT HOURS SINCE LAST CHECK	9. DAYS SINCE LAST CHECK	
	13 MAY 1957	4453	3	5	236	PNS	CALENDAR ODD	104	82	
B. ENGINE HISTORY	1. ENGINE MODEL	2. ENGINE SERIAL NUMBER	3. FLIGHT HRS. SINCE ACCEPTANCE	4. NUMBER OF OVERHAULS	5. WAS DIR. REQUESTED?	6. FLT. HRS SINCE LAST OVERHAUL	7. LAST OVERHAUL ACTIVITY	8. TYPE OF LAST CHECK PERFORMED	9. FLIGHT HOURS SINCE LAST CHECK	10. DAYS SINCE LAST CHECK
	(1) R1820-86A	520333	3470	5	YES	134	PNS	CALENDAR ODD	104	82
	(2)									
	(3)									
C. COMPONENT HISTORY	1. COMPONENT INVOLVED NOMENCLATURE	2. MANUFACTURER'S PART NUMBER	3. TOTAL HRS. ON PART	4. NO. OF OV-HAULS	5. HOURS SINCE LAST OVERHAUL	6. OVERHAUL ACTIVITY	7. WAS DIR. REQUESTED?	8. SER. NO. FJB/ANPELUP		
	(1)									
	(2)									
	(3)									
D. INCIDENTS & GROUND ACCIDENTS#	1. PARTS REPAIRED		3. DIRECT MANHOURS INVOLVED		2. PARTS REPLACED					
	PART NUMBER	NOMENCLATURE			PART NUMBER	NOMENCLATURE				
E. ENGINE FAILURES	JET ENGINE FLAMEOUT (Include intentional securing to prevent engine damage)									
	AT TIME OF FLAMEOUT	1. ALTITUDE	2. MS	3. RPM	4. EGT	5. MANEUVER AT TIME OF FLAMEOUT	6. FUEL FLOW	7. ALTITUDE		
	8. G FORCES	9. RELIGHT <input type="checkbox"/> ATTEMPTED <input type="checkbox"/> ACCOMPLISHED	10. ALTITUDE	11. MS	12. MAX EGT	13. FUEL CONTROL <input type="checkbox"/> PRIMARY <input type="checkbox"/> MANUAL	14. NO. RELIGHT ATTEMPTS			
	INTENTIONAL SECURE	15. ENGINE SYMPTOMS	16. CAUSE OF SYMPTOMS							
	RECIPROCATING ENGINE FAILURE									
	17. ALTITUDE	18. MS	19. ALTITUDE	20. RPM	21. MAP	22. TORQUE/BNHP	23. FUEL FLOW PRESSURE	24. OIL PRESSURE		
	INTENTIONAL SECURE	25. ENGINE SYMPTOMS	26. CAUSE OF SYMPTOMS							
	IDENTIFY OTHER REPORTS CONCERNING THIS mishap									
F. OTHER REPORT	1. AMPFUR SERIAL NUMBER									
	2. DIR MESSAGE REQUEST DATE-TIME-GROUP 092001 8 AUG 67									
	3. OTHER PRELIMINARY MESSAGE DTG 090650 8 AUG 67									
	4. SUPPLEMENTARY MESSAGE DTG 092358 8 AUG 67 DIR REPLY: NAVAIREWORKPAC PNCLA 221956Z AUG 67									

OPNAV FORM 3750-1A (Rev. 3-63)

SPECIAL HANDLING REQUIRED in accordance with
Para. 64, OPNAV INSTRUCTION 3750.1, (latest edition)

OPNAV REPORT 3750-1

G. SHIPS DATA

WEATHER/H. DEPLOYMENT

H. DEPLOYMENT		FOR ACCIDENTS ABOARD CARRIERS (complete on pilot)			
1. DATE DEPLOYED COMUS	3. DAY HOURS/LANDINGS SINCE DEPLOYMENT	4. DAY HOURS/LANDING LAST 30 DAYS			
2. NO. DAYS OPERATING PERIOD					
5. INST. HOURS LOGGED SINCE DEPLOYMENT ACTUAL/SIMULATED	6. NIGHT HOURS/LANDINGS SINCE DEPLOYMENT	7. NIGHT HOURS/LANDINGS LAST 30 DAYS			
WEATHER AT SCENE OF MISHAP					
1. CEILING	2. VISIBILITY	3. RELATIVE WIND DIRECTION AND VELOCITY	4. TEMPERATURE SURFACE OUTSIDE AIR	5. DEW POINT	6. ALTIMETER SETTING
12,000	5	100° 3 KNOTS	82	76	29.98
7. OTHER WEATHER CONDITIONS (winds aloft, icing level, sea state, density altitude, as appropriate)					

PART III ADDITIONAL INFORMATION

PART	SECTION	ITEM	1.	REMARKS	2. COPY DISTRIBUTION
					700 NAVJNSAFECN DIRECT (JAG) 1CC NAVATRSYS.COM HQ 1CC CNAB/TRA 1CC CNATRA 1CC NAVPLANTREPO COLUMBUS 1CC DIRECTOR AFIP 1CC TRARON TWO FILES
COST DAMAGE TO:			3. GOVERNMENT PROPERTY	4. PRIVATE PROPERTY	5. DATE SUBMITTED TO GO
			NONE	\$400 (EST)	28 AUGUST 1962

PART IV SIGNATURES OF THE BOARD

1. SUP (b) (6)	2. PER (b) (6)
(b) (6) LCDR TRNG ADMIN UNIT GILLET	(b) (6) LCDR MAJLT. CONTROL UNIT GILLET
(b) (6)	4 (b) (6) (b) (6) LT AVIATION SAFETY UNIT GILLET

* When preparing Incident and Ground Accident reports, items indicated by an asterisk in the upper right hand corner must be filled in. Other items considered appropriate should also be filled in.

THE ACCOUNT

PART V THE ACCIDENT

T-28C, Buno 140655, side number 2G-274 was scheduled for a Night Navigation Two hop, a VFR round-robin, on the night of 8 August 1967. The pilot, Ens Mark R. Kaufman, along with the other students were briefed for their flight by Lt (b) (6) (Enclosure 8). The brief covered taxi, take-off and climbout instructions in accordance with T-28 Flight Training Instructions for night flying (CNAFT P-277 Rev. 4-67).

The Alfa route was to be flown as follows: Whiting to Andalusia (A-1) to Bay Minette (A-2), to Whiting, to Greenville (A-3), to Bay Minette (A-4), to Whiting (destination), for a total time of 2.5 hours. The runway in use was runway 31, requiring a right turn after take-off of approximately 60° to the on-course heading.

Pre-flight, turn up and taxi was normal. The launch was commenced at 2231 R. Three night space pilots (instructors who check student's aircraft over each point with call signs of A-1, A-2, etc.) were launched in 2G-306, 2G-310, and 2G-301. The first student in 2G-272 was then launched followed by the fourth space pilot in 2G-309 (Enclosure 9). The second student was launched at 2237 R. 2G-274 reported his second magneto check satisfactory and take-off check list complete (Enclosure 10). He was then given a flashing green light by the Runway Duty Officer clearing him onto the duty runway. After a two minute interval was established 2G-274 was given a steady green light by the RDO clearing him for take-off. 2G-274's take-off was normal at 2239 R and a climb to enroute altitude was commenced. After reaching approximately 500 feet

altitude, the aircraft commenced a right turn. The nose was observed to fall and the aircraft entered a diving right turn (Enclosures 11, 12, 13). Contact with the ground was made at 2240R and the aircraft exploded on impact. First contact with a stationary object occurred when the aircraft brushed through the top of a pecan tree 237 feet prior to impact with the ground (Enclosure 1). First ground impact was made with the stbd wing tip. The aircraft attitude at this time is estimated to be 30° right bank and 20° nose down, airspeed at 200 knots, with a heading of 055° (Point A, Enclosures 2 and 3). The aircraft nose began a slight movement to the right and the wing attitude started a rotation from right to left. The nose impacted 82 feet beyond the initial point of touchdown, scooped out a trough 19 feet long and 18" deep (Point B, Enclosures 2 and 3). The nose was then forced up into the air. The engine separated from the airframe at this point and the initial fire began. The aircraft heading at this point was approximately 095° with the path of travel still at 055°. The wing rotation (right to left) continued. The airframe hit three medium size stumps and the stbd wing struck a tree while it traveled at approximately 2 feet above the ground (Point C, Enclosures 2 and 3). It began a tail downward movement and struck the ground again with the port wing tip and port horizontal stabilizer tip (Point E, Enclosures 2 and 3). At this time the aircraft began an over the port wingtip movement while dragging the tail section causing the aircraft nose to move back toward the flight path direction and rolling into an inverted attitude (Point F, Enclosures 2 and 3).

The airframe impacted a third time in the inverted position, nose

down and wings level (Point G, Enclosures 2 and 3). The heading was approximately 20° to the left of the flight path. The impact flattened the cockpit windscreen and smashed the canopy. Shortly after this impact the leading edge of the port wing stub dug into the ground causing the aircraft tail to flip up into the air (Scene F, Enclosure 3). The fuselage came to rest right side up on a heading of 150° relative to the flight path (Point J, Enclosures 2 and 3).

After separating from the airframe the engine travelled in the air for 241 feet and impacted a second time. After the second impact the engine rolled on it's cylinders for an additional 499 feet coming to rest 338 feet beyond the fuselage (Enclosure 2).

Tower personnel, the RDO, and the night space pilots were alerted to the crash by the explosion when the nose impacted (Enclosures 15 and 16). Crash vehicles were alerted and sent to the crash site (Enclosure 17 and 18). The crash vehicles and rescue personnel extinguished the fire in the fuselage upon arrival (Enclosure 19). The pilot's body was removed when the flight surgeon arrived. The secondary fires were then extinguished. Upon determination of a crash the night navigation launch was cancelled and the other airborne aircraft were landed at South Whiting Field without further event.

PART VI. DAMAGE TO AIRCRAFT

The aircraft received Alfa damage as a result of the crash and fire. Both outer wing panels and the outer five feet of each wing were torn from the aircraft (Enclosure 5). The starboard wing received major

damage on initial impact, with the outer part of the wing being curled up from the tip while absorbing the shock of the impact. The starboard outer wing panel and outer portion of the wing separated from the aircraft when it impacted in the inverted position (scene G Enclosure 3). The port wing received major damage on the second impact. The aircraft hit on the wingtip folding the aileron under the wing, then tearing the outer wing panel off the aircraft. The wing sustained further damage as the airframe vaulted over the wing. The port wing stub also dug into the ground on the third impact.

The empennage section was torn, dented and scraped throughout but remained relatively intact, and undeformed with the exception of the port horizontal stabilizer. The port horizontal stabilizer was curled under from the tip on the second impact. The entire section with the exception of the control cables was separated from the fuselage as the aft part of the fuselage was twisted on the second impact (Enclosure 7).

The fuselage forward of the rear cockpit was destroyed first by impact and then by fire (Enclosure 6). At the initial nose impact the propeller broke off the engine and the engine separated from the fuselage. Most of the cowling pieces went with the engine and were strewn along the engine's path. The fuselage was dented along the port side when the aircraft hit a fence and three stumps. The empennage section was torn loose from the torque forces developed as the port horizontal stabilizer was curled under (Enclosure 7). The top forward portion of the aircraft and the canopy received major damage on the third impact as the aircraft hit inverted about 20° nose low. The windscreen and canopy were smashed

and shattered. The nosewheel well separated at this point and was thrown forward. Damage sustained on the final impact is considered minimal except for excessive "g" loads experienced as the aircraft hit. This impact caused the instrument panels to break loose, the landing gear to slip from the uplocks, and the aft canopy frame to break away from the fuselage.

The fire first started on nose impact. The explosion that occurred was primarily burning fuel from the starboard wing. The fire followed the aircraft to its final position where it consumed the entire front cockpit area reducing it to ashes except for heavy steel parts. The engine and cowling did not catch fire or show evidence of burning (Enclosure 4).

PART VII THE INVESTIGATION AND ANALYSIS

(b) (5)

The aircraft first struck the top 18 inches of a tree that was approximately 85 feet tall. 237 feet later the starboard wing tip touched down indicating a 20° dive angle. The angle of bank at this touchdown is estimated at 30°. Parts from the aircraft indicated only the manner in which the aircraft broke up on its successive impacts.

(b) (5)

The investigation of the cockpit area revealed the following:

1. The pilot's shoulder harness lock was in the unlocked position.

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SPECIAL HANDLING REQUIRED IN ACCORDANCE WITH OPNAVINST 3750.6 SERIES

2. The canopy handle was in the normal open position (rear cockpit), but the canopy remained closed.
3. The magneto switch was near the off position (rear cockpit).
4. The flap handle was in the down position (rear cockpit).
5. All other switches were in their normal positions.

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PAGE 14 OF 15 PAGES
SPECIAL HANDLING REQUIRED IN ACCORDANCE WITH OPNAVINST 3750.6 SERIES

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PART VIII CONCLUSIONS

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PART IX RECOMMENDATIONS

(b) (5)

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1. OVERHAUL ACTIVITY NAVAIREWORKFAC PNCLA		2. REPORT NO. 668	3. DATE OF D/I 8-21-67	4. ASSEMBLY NOMENCLATURE AND PART NO. MAGNETO MIS-9F-301		ENGINE <input type="checkbox"/>
5. ASSEMBLY (Model) S9LU-6		6. ASSEMBLY (Serial) 2G6307	7. ASSEMBLY MFR 01843	8. DATE REMOVED UNKNOWN	9. REMOVED FROM (Eng No) R1820-86A	10. REMOVED FROM (Eng Ser) R1520333
11. TOTAL HRS SINCE NEW	12. HRS SINCE LAST O/M 131	13. DATE LAST O/M AUG-66	14. LAST OVERHAUL ACTIVITY NAVAIREWORKFAC PNCLA		15. NO. PREV O/M'S	16. AIRCRAFT (Model) T-28C
17. AIRCRAFT (SERIAL) 140655						
18. OPERATING ACTIVITY VT-2		19. PUR-EPS-AAR-1/77H/DA AAR 1-68A	20. REASON FOR REMOVAL AND CODE Accident/Incident Damage 4B			
21. FINDINGS <input checked="" type="checkbox"/> NO DISCREPANCY		22. BASIC (MFG/DESIGN) DISCREPANCY <input type="checkbox"/>		23. NON-BASIC (MAINT/OPER) DISCREPANCY <input type="checkbox"/>		24. FOREIGN OBJECT DAMAGE <input type="checkbox"/>
25. DESCRIPTION OF FINDINGS (Include name and part no. of primary part failure)						
<p>*Accessory Card not available. NASCRPNCLA CONTROL NR R1820-8-68</p> <p>COPY TO: NASC (AIR-4113) " (AIR-5362) " (AIR-4041) NAVAIRSYSOMREP PNCLA NASTECHREP WOODRIDGE NAVAVNSAFECEN ✓</p> <p>NATSF PHILA CNATRA CNABATRA NAAS WHITING FLD IMA TRARON TWO PLANNING-MATERIAL</p>						
26. DISCREPANT PARTS (Part No.) COND.						
27. PERTINENT BULLETIN, CHANGES, ETC... INCORPORATED						
28. NUMBER YES NO						
29. CONCLUSIONS						
M a g n e t o was operating satisfactorily prior to accident.						
30. RECOMMENDATIONS						
PRIORITY						
31. REQUESTED BY <input checked="" type="checkbox"/> PRIORITY DIR NASCR PNCLA		32. REFERENCE MSG 101424Z Aug 67		33. GRES 114 APPLICABLE INCORPORATED		34. DATE 28 Aug 1967
35. SIGNATURE H. YESNES		36. TITLE Weapons Engineering Department Head				
DISASSEMBLY AND INSPECTION REPORT NAVMPS FORM 4730/2 (11-61)						REPORT SYMBOL BUMPS 4730-2

PRIORITY

REPORT SYMBOL SUMEP-6730-2

708081--

1. OVERHAUL ACTIVITY		2. REPORT NO.	3. DATE OF D/I	4. ASSEMBLY NOMENCLATURE AND PART NO.		(163)	ENGINE																																	
NAVAIREWORKFAC Pensacola		663	8/21/67	Propeller																																				
5. ASSEMBLY (Model)	6. ASSEMBLY (Serial)	7. ASSEMBLY MFR	8. DATE REMOVED	9. REMOVED FROM (Eng Mod)	10. REMOVED FROM (Eng Ser)																																			
43D50-321	202029	73030		R-1820-86A	BL520333																																			
11. TOTAL HRS SINCE NEW	12. HRS SINCE LAST D/I	13. DATE LAST D/I	14. LAST OVERHAUL ACTIVITY		15. NO. PREV D/I'S	16. AIRCRAFT (Model)	17. AIRCRAFT (SERIAL)																																	
2684	130	4/8/67	NAVAIREWORKFAC Pensacola		3	T-28																																		
18. OPERATING ACTIVITY		19. PUR-EPS - AAR - I/PN/SA	20. REASON FOR REMOVAL AND CODE																																					
TRARON TWO		AAR-1-68A	Accident/Incident Damage 4B																																					
21. FINDINGS		22. DISCREPANCY		23. DISCREPANCY		24. DISCREPANCY																																		
<input type="checkbox"/> NO DISCREPANCY <input checked="" type="checkbox"/> BASIC (MFG/DESIGN) DISCREPANCY <input type="checkbox"/> NON-BASIC (MAINT/OPER) DISCREPANCY <input type="checkbox"/> FOREIGN OBJECT DAMAGE																																								
25. DESCRIPTION OF FINDINGS (Include name and part no. of primary part failure)																																								
NAVAIRSYSCOMREP PNCLA CONTROL NUMBER R1820-8-68 REFERS TO: NATSF PHILA CC: NAVAIRSYSCOM (AIR-4105B) NAVAIRSYSCOM (AIR-4041) NAVAIRSYSCOM (AIR-4113) NAVAIRSYSCOMREP PNCLA NAVAVNSAFECEN CNABATRA CHATRA NAVAIRSYSCOMTECHREP WOODRIDGE TRARON TWO (AVSAF OFFICER) TRARON TWO (A/C MAINT OFFICER) NAVPLANTREPO, NAA COLUMBUS DCASO, WINDSOR LOCKS NAS PNCLA																																								
26. PERTINENT BULLETIN, CHANGES, ETC., INCORPORATED																																								
<table border="1"> <thead> <tr> <th>NUMBER</th> <th>YES</th> <th>NO</th> </tr> </thead> <tbody> <tr><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td></tr> </tbody> </table>								NUMBER	YES	NO																														
NUMBER	YES	NO																																						
27. CONCLUSIONS																																								
See below.																																								
28. RECOMMENDATIONS																																								
None.																																								
29. REQUESTED BY		REFERENCE		30. SIGNATURE		31. DATE																																		
NAVIRSYSCOMREP PNCLA		101424Z AUG 67		H. YESNES		24 Aug 1967																																		
32. PRIORITY		33. TITLE		34. DISASSEMBLY AND INSPECTION REPORT		35. REPORT SYMBOL																																		
H. YESNES		HEAD WEAPONS ENGINEERING DEPT.		NAVWPS FORM 4730/2 (11-61)		REPORT SYMBOL BUNWPS 4730-2																																		

PRIORITY

(b) (5)

26. FINDINGS:

(b) (5)

SUPPLEMENTARY REPORT

USNAAS, WHITING FIELD, MILTON, FLORIDA, T-28, BUNO 140655,

8 August 1967, KAUFMAN, Mark Richard.

R 11-9-67
82